



Latest update on J1 VIIRS geolocation

NASA VIIRS Characterization Support Team (VCST) Geometric Calibration Group

Guoqing (Gary) Lin, SSAI/GSFC Code 619 Robert E. Wolfe, NASA/GSFC Code 619 John Dellomo & Blanche Pfarr, GST/GSFC Code 619 Ping Zhang & Bin Tan, SSAI/GSFC Code 619 James C. Tilton, NASA/GSFC Code 606

NASA Ocean Biology Processing Group (OBPG) Fred Patt, SAIC/GSFC Code 616

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- Thanks to the JAM, Rosalie Marley, for helping us resolving DRs in the DPE/AMP/GRAVITE.
- Thanks to NOAA JPSS MOT, NASA FDF, BATC for assistance in understanding the SNPP orbit dynamics and attitude issues.
- And thanks to other members of the JPSS family....







1. J1 updates from SNPP VIIRS geolocation

- Mounting matrix & possible on-orbit correction
- Differences from SNPP
 - LSF/FOV, BBR, focal length, scan rate
 - DNB thanks to STAR colleagues for code update

2. SNPP updates for J1 VIIRS geolocation

- Improvements completed

– Improvements in the making

3. Concluding Remarks

• Extra: Posters 1) J2 VIIRS test data analysis results

2) On-Orbit Measurement of the Focal Length of the SNPP VIIRS Instrument





Part 1: J1 updates from SNPP VIIRS geolocation

- Mounting matrix & possible on-orbit correction
- Differences from SNPP
 - Optical performance, BBR
 - Focal length, scan rate, scan-to-scan underlaps
 - DNB geolocation with options in reduced aggregation mode – thanks to STAR colleagues for code update



J1 Post-vibe, PreTV data from BATC report

IMF Delta from Average (arcsec)			+0G				
	Bias	VIIRS	ATMS	CrIS	CERES	OMPS	Average-bias
θХ		17.5	-94.1	47.9	10.9	18.0	49
θY		41.6	-11.0	66.0	-70.5	-27.8	-600
θZ	-50	-51.8	180.6	3.1	-260.6	-123.4	-202

Mounting coef T_inst2sc =
0.99999997 0.00019698 0.00015131
-0.00019698 0.99999998 -0.00005173
-0.00015132 0.00005170 0.99999999

Equivalent to Euler angle correction (arcsec)

Angle	At-launch	Measured Delta		
Yaw	40.6, postTV?	?		
Roll	-10.7, postTV?	?		
Pitch	-31.2, postTV?	?		

Δscan 1,118 m

SNPP initial	Correction to Instrument to Spacecraft Alignment (arcsec)	Angle	At-launch	Measured Delta
on-orbit		Yaw	33.2	62.2
geolocation		Roll	41.2	-268.5
correction		Pitch	-59.3	212.5
			Nadir equiva Δtrack -775	alent: 5 m,



1.4

1.3

Spatial Response (Area_under = 1)

Position (Fraction of Non-Aggregation Pixel) Position (Fraction of Non-Aggregated Pixel) > J1 VIIRS has improved optical performance over SNPP, which

shows up in band I1 LSFs more prominently





J1 M01D08

LSF = Line Spread Function

1.4

1.3

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1.5

J1I01D16





J1 VIIRS has better focus while SNPP VIIRS has de-focus in VisNIR bands
 a "trade" in focal length : scan rate : scan-to-scan underlap
 I-bands under-sample while M-bands over-sample the earth in the un-agg zones









J1 BBR performance is better than SNPP BBR in the scan direction

In the track direction, J1 Bands on cold FPAs shifted ~ 50 m from bands on VisNIR FPA.

Additional geolocation monitoring of band I3 is planned to monitor the shift







	EFL (mm)	Scan rate (rad/s)	Scan period (s)	EV scan angle (deg)	EV ground distance (km)
SNPP	1135	3.531	1.7793	+/- 56.28	+/- 1528
J1	1141	3.517	1.7867	+/- 56.04	+/- 1510

- Focal length affects scan rate per BBR requirements
- SNPP focal length is too short so that the scan rate is maxed out (+0.4%)
 - On-orbit measurement for SNPP EFL is ~ 0.5% shorter than nominal, see poster
- J1 focal length is nominal, which will be assessed onorbit



11.75 -75 -60 -45 -30 0 15 30 45 60 75 90 50 100 150 SSP Latitude (deg) Distance from nadir (km)

- Largest underlap occurs at nadir and ~15° N in the ground track
- Higher terrain opens up underlap more, ~ 14 m/km terrain height
- Spacecraft jitter makes underlap more in one scan but less in the next

2. My eRooms > JPSS Science > J3-J4 VIIRS Performance-Improvement > JPSS Orbit Parameters and Variations

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Nadir ground dist (km)

12.2

12.1

11.9

11.8

11.7 11.6

-90

12

overlap

12.05 for a sample orbit 2014-10-30 00:31:49 -- 02:13:19z 12.4 Field of regard 12.00 12.3

J1 VIIRS built to nominal optical performance results in \bullet unexpected scan-to-scan underlap^{1,2}







^{1.} Lin, G., R.E. Wolfe, J. C. Tilton, (2016), "Trending of SNPP ephemeris and its implications on VIIRS geometric performance", Proc. of SPIE, Vol. 9972, 99721K, doi: 10.1117/12.2239043.



J1 DNB aggregation mode change



-- Thanks to STAR colleagues for code update



- SNPP baselined pixel size is ~ 750 m
- J1 "Option21" has pixel size growing larger to the end of scan
- Geolocation is extrapolated (no encoder data) post-nadir for scan angle > 56.5°

We will assist validating J1 DNB on-orbit geolocation





Part 2: SNPP updates for J1 VIIRS geolocation

- Improvements completed for SNPP \rightarrow J1
 - Ground-truthing tool, the Control Point Matching (CPM)
 - Timely delivery of spacecraft diary data
 - One-sided star tracker re-alignment - no more
 - SNPP star tracker cooling - reduced frequency of attitude excursions
 - Leap second insertion
- Improvements in the making for SNPP \rightarrow J1
 - Position error after inclination adjust maneuvers
 - Imagery EDRs using ellipsoidal vs TC geolocation input



CPM code update





- Bugs are fixed in handling the cross-aggregation zone and cross-scan boundaries (& fixed a few other bugs)
- Measurements of biases are about the same -- drifts in the track (pitch) direction

Timely spacecraft diary delivery





- Large circles for control spec outage; Small dots hint knowledge spec outage
- Star tracker cooling improved SNPP attitude performance
- We are seeking for further improvements³
- We are developing SW with Kalman filter to refine the attitude for NASA SIPSs
- J1 is expected to perform better but we need to monitor/verify

3. My eRooms > S-NPP Flight Operations and Support > FARB > All Discussion Topics--Artifacts and Minutes > DR 6348--SNPP STAR TRACKER

Impact from leap second insertion



Original

Leap second insertion at the start of **1 July 2012** impacted geolocation for ~ **3 hours**, primarily due to convergent time in shifted position of ~ 7 km.

Improved

Leap second insertion at the start of **1 July 2015** and later events impacted geolocation only **1-min**, with ephemeris reinitialized to GPS (command ADICVGPS)





Part 2: SNPP updates for J1 VIIRS geolocation

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Minimize errors after IAMs





- An Inclination Adjust Maneuver (IAM) pushes the orbit sideway, making the orbit plane to incline toward earth rotation axis (once per year)
- The on-board orbit propagator does not respond to the position change quick enough, yielding position error up to **3000 m** in this case
- Orbit knowledge convergence takes about 1.5 to 2 orbits.
- The MOT folks "are looking to into being able to" re-initialize ephemeris with GPS with command ADICVGPS for IAM planned on 9/20/2017
- The same could be applied to DMUs/RMMs (~ 5 times per year), that have position errors ~ 150 m

Assisting Imagery EDR Team: Terrain To be or not to be corrected?





- The error in un-corrected ellipsoid geolocation data products depends on position off-nadir and terrain height
- Maximum error factor is ~2.7 km / 1 km terrain height
- Imagery EDR is currently based on ellipsoid geolocation – mountains move from one swath to the next

Note: terrain corrected geolocation products are available for IMG, MOD and DNB





Concluding Remarks

- J1 VIIRS geolocation is expected to perform better than that of SNPP VIIRS, as anomalies (issues) have been resolved, caveats (known deficiencies, challenges and concerns, and curiosities) have been looked into, and lessons learned
 - Due diligence still needs to be performed
- J1 on-orbit geolocation Cal/Val preparation is well underway

Posters

- 1) J2 VIIRS ground test data analysis
- 2) On-Orbit Measurement of the Focal Length of the SNPP VIIRS Instrument





Backup

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VCST/GEO 21



Control point chips supplements





Over 1200 Ground Control Point (GCP) chips of Landsat TM red band (0.64 µm) 30 m nadir resolution, 2003 or earlier. Some chips are out-of-date and are removed from CPM for VIIRS. To add ~4000 (planned) Ground Control Point (GCP) chips of Landsat-8 30 m nadir resolution. L8 OLI red band 4 (0.64 μ m) = VIIRS I1 . Data for all other 10 bands are available. Priority is L8 TIRS band 6 (1.6 μ m) = VIIRS I3. Both coastal and inland chips are being acquired.



Improvements & progresses by addressing anomalies, caveats & curiosities



- Rare cases of large errors may not show up well in the mean value
- They do affect "spread" (standard deviation) and uncertainty (RMSE = Root Mean Square Error)
- Specification in "3σ" may be violated when we interpret it as 99.73% of occurrences or detectible measurements